# SIMPLIFYING OPERATIONS WITH AN UPLINK/DOWNLINK INTEGRATION TOOLKIT

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#### ABSTR ACT

The Operations Engineering 12113 (01 3L) at JPL has developed a simple, generic toolkit to integrate the uplink/downlink processes, (often called *closing the loop*,), in JPI,'s Multimission Ground Data System. This toolkit provides capabilities for integrating telemetry verification points with predicted spacecraft commands and ground events in the Mission Sequence Of Events (SOE) (1 ocument. In the JPL ground data system, the uplink processing functions and the downlink processing functions are separate subsystems that are not well integrated because of the nature of planetary missions with large one-way 1 ight time.s for spacecraftto-ground communication. Our new closedloop monitoring toolallows an analyst or mission controller to view and save uplink commands and ground events with their corresponding downlinked telemetry values regardless of the delay indownlink telemetry and without requiring mal-time intervention by the user.

An SOE document is a time-mdcmxl list of all the planned ground and spacecraft events, including all commands, sequence loads, ground events, significant mission activities, spacecraft status, and resource allocations. The SOE document is generated by expansion and integration of space.cmft sequence files, ground station allocations, navigation file.s, and other ground event files. This SOE generation process has been automated within the OEL and includes a

graphical, object-oriented SOE editor and real-time viewing tool running under X/Motif. The SOE toolkit was used as the framework for the integrated implementation.

The SOE is used by flight engineers to coordinate their operations tasks, serving as a predict data set in ground operations and mission control. The closed-loc)p SOE toolkit allows simple, automated integration of predicted uplink events with correlated telemetry points in a single SOE document for on-screen viewing and archiving. It automatically interfaces with existing real-time or non real-time sources of information, to display actual values from the telemetry data Stream.

This toolkit was designed to greatly simplify the user's ability to access and view telemetry data, and also provide, a means to view this data in the context of the commands and ground events that are used to interpret it. A closed-loop system can prove especially useful in small missions with limited resources requiring automated monitoring tools. This paper will discuss the toolkit implementation, including design trade.-offs and future plans for enhancing the automated capabilities.

#### INTRODUCTION

The Operations I ingineering 1 ab (01;].) at NASA's Jet l'repulsion Laboratory has developed a simple, generic toolkit that

integrates uplink events with downlink telemetry information, (often called *closing* the loop,). This toolkit provides capabilities for integrating telemetry verification points and ground monitoring information with planned spacecraft commands and ground events in the Mission Sequence Of Events (SOE) schedule. in the existing SOE for planetary missions, each spacecraft command item has a descriptive text field that contains a list of related engineering telemetry parameters. These parameters are monitored by mission controllers using a data monitor tool that processes and displays the downlink telemetry stream. However, the relevant downlink telemetry for command verification may not arrive for hours or even days after the commands have been sent because of the large one-way light times and limited contact periods of some planetary missions. The closed-loop system will integrate these tasks by interfacing the SOE with the mal-lime. or non-real-time telemetry data streams and automatically append appropriate channel values and limit checks with command and ground event items. Our new closed-loop monitoring tool allows an analyst or mission controller to browse and archive uplink commands and ground events with their corresponding downlinked telemetry values regardless of the delay indownlink telemetry and without requiring mal-time intervention by the user. 1 figure 1 shows a sample SOE with integrated telemetry channel information.

#### BACKGROUND

An SOE document is a time-mdcrcd list of all the planned ground and spacecraft events, including all commands, sequence loads, ground events, significant mission activities, status and resource allocations. The SOE document is generated by the multimission control team by expansion and integration of sequence and ground files. This SOE generation process has been automated and includes a graphical, ohic.ct-oriented SOE editor and viewing tool running under X/Motif. The SOE is used by the mission controllers and spacecraft and instrument engineers to coordinate their operations tasks, serving as a predict data set in ground operations and mission control.

#### APPROACH

The Operations Engineering Lab (OEL) proposed a research task to design and implement a toolkit that allows simple, automated integration of predicted uplink events with actual downlink telemetry in a single SOE document for viewing and archiving. This integrated SOE serves as the basis for a closed-loop monitoring toolkit that can automatically interface with existing real-time or non-real-time sources of information and display only selected values from the telemetry data stream.

A significant research effort was in the design and implementation of the interprocess communication interfaces and interactive controls for retrieving and passing information from a variety of downlink processing applications to the SOE tool. A flexible approach was chosen to allow phasing, of planned future enhancements, including expansion of the SOE capabilities for automated mission controller logs, telemetry logging, system test procedure execution, and automated command verification.

During the requirements analysis, the JPL Multimission Control Team (MCT) indicated that the closed-looJ> monitoring system should include automated log keeping capabilities for mission controllers in order to include their real-time Jogs in the sequence of in the current operations c.vents. environment, the MCT logs real-time information on specific uplink and downlink events on hand-written forms. The MCT cannot meet an electron ic logging requirement without automation tools - manually typing a 1 o g report would be impossible during intense periods. Although the SOE tool can currently be used to enter comments and act as a logging tool, the mission controller logging requirements were very broad and not clearly defined enough to implement within an already-developed tool such as the SOE editor. '1'bus, we decided to implement a separate tool for automated logging that would integrate manual log inputs, predicted events input files (such as the SOE), real-time broadcast data, and out put from other

downlink telemetry processing and monitoring applications.

#### OEL LOGGING TOOL.

The OEL Logging Tool (OLOG) provides automated and manual logging of predicted and actual mission events in a graphical easy-to-use format. The user interface and pull-down menus are completely configurable by an individual end-user to meet mission-specific needs. An initialization file can be read on startup that customizes menu options and defaults. The OEL Logging '1'001 is shown in Figure 2.

The OEL Log Tool provides capabilities for manual entries to the log, either by allowing a user to select items offpre-defined menus or entering text manually into the entry area. The pm-defined menus and default values can be customized for various teams by creating an initial ization file. The OEL Log Toolis also designed to interface in mal-time with external input sources (downlink telemetry, monitor data, interprocess messages, predicts and actuals files), thus providing capabilities for automated log entries. The tool is designed to allow automated and manual log entries to occur concurrently with automatic ordering based on time tags. The current tool implementation includes several mechanisms for communicating with downlink processes basal the MGDS custom Data Transport Subsystem (1)'1'S) services. We have also implemented a telemetry data processing program that captures and processes data from the |cal-time telemetry data stream or spooler files, and passes log messages into the OELLog T'001.

#### INTERPROCESS COMMUNICATION

The OEL Log Toolusesthe MGDS 1)'1'S services to implement message passing capabilities. The tool connects to a broadcast channel or virtual circuit if a channel option is chosen. While processing input events, the tool continually checks for pending messages.

Several programs and routines have been written to implement real-time downlink data processing functions and cooperative

message passing mechanisms. These programs provide a flexible approach to process various input data sources, which are then reformatted as log messages to be sent to the OEI, 1.og '1'001.

- A program that uses a DTS broadcast channel or virtual circuit for sending timestamped messages. TheOELLog '1'001 cooperates by receiving and processing the messages for input into the log in time order.
- A generic routine that uses the MGDS Data Transport functions (DTS) to connect to a broadcast channel or virtual circuit for sending lime-stamped messages and a corresponding OBL Log Tool routine that creates a broadcast channel or virtual circuit to receive messages.
- A program that reads the output from an existing real-time Smart Alai.IN Tool.It processes the output and then sends log messages to the ()] I. Log '] '00].
- program that reads MGDS downlinked telemetry data from a virtual circuit, real-time broadcast channel, a bytestream file, and/or a spooler file using the DTS functions. It p rocesses the data by parsing telemetry headers and checking for selected data types which are then Cxtl'acted> processed, and passed to the OEL Log Tool, It uses DTS services to send log messages to the OELLog Tool. This program provides a model for building additional programs for processing data from real-time telemetry streams or from data output by other applications and then generating log messages to be integrated with the predict in formation in the log tool.

Figure 3 shows the OEL Log Tool and its telemetry processing interfaces.

#### CL0s1 D-LOOP sol TOOL

The new closed-loop SOE too] is designed to allow integration of uplink predict events with actual downlink telemetry values.

Ground event and spacecraft command items in the SOE have associated telemetry parameters that can be used to verify the execution of each command or ground event or to establish the state of the spacecraft or ground system. These parameters are IIOW interleaved with the commands and ground events in the closed-loop SOE, for easy access and viewing by the analyst. The Closed-loop monitoring system can automatically interface with existing real-time. sources of information, to capture and display actual values from the telemetry data stream. The graphical SOE viewing tool allows a user to highlight events of interest with a mouse on the screen and mark them for automatic alarm notification in real-time. This tool also allows a user to run the SOE in demandmode, specifying any time desired, and to search or Step through the document for events of interest.

The following capabilities have, been incorporated into the SOE tool to implement real-time, closed-loop monitoring capabilities:

- Real-Time Capabilities: A scroll bar has been added to allow for scrolling during real-time viewing. Real-time viewing can be accessed and controlled from a menu. It is possible to step through each event under user control or to runthe SOE in md-lime or in playback mode. The real-time control capabilities are based on an interrupt algorithm.
- Channel Objects: A new item format has been defined for the SOE to allow creation of a telemetry channel object. Each channel object is defined by a description field and is associated with the uplink predict item preceding it. "I'here can be multiple channel objects associated with a single uplink e v e n t. The uplink/downlink command/channel integ ration is based on cross-referenced information within the telemetry and command dictionaries.
- Channel Options: A new option but ton has been implemented that allows a user to optionally display channel records in the SOE.
- Downlink Interface: Interprocess communication and data transport functions have been implemented to allow

integration of the downlink telemetry stream.

Figure 4 shows the graphical closed-loop SOE tool.

#### BENEFITS

New missions are demanding electronic mission controller logs rather than the current han(i-written reports. Without the automated log keeping, capabilities in the new OELL og '1'001, a controller would spend most of their time manually typing in electronic log reports.

The closed-loop SOE and OEL Log Tool implementations have laid the groundwork for an advanced closed-loop monitoring system that can significantly reduce the need for operations teams to understand the complex set of processing and display tools in the existing JPL Multi mission Ground Data System (MGDS). The current MGDS approach requires understanding multiple subsystems, and their often subtle interfaces, to allow an end-to-end processing of downlink data. There is also no current method to integrate or compare predict with actual values. The closed-loop SOE tool can greatly simplify a user's ability to access and view telemetry data, and provides a means to view this data in the context of the commands and predicted values that are used to interpret it. In this context, it is expected that significant cost savings can be realized from the productivity improvements that will be realized over the hundreds of current users of the SOI 3 document and related uplink tools.

Some proposed areas of future work include expansion of the closed-loop SOE capabilities for adaptation to system test, dynamic stams based on tolerances from predicted events, automated monitoring of spacecraft and ground system configurations at selected times, mal-time timeline display of the SOE, generation of as-flown SOE schedules, and automated command verification. An advanced closed-loop monitoring system is essential to a more automated monitor and control system and to significant productivity improvements for the smaller missions of the future.

### A(:KNOW1,10 )GMENTS

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T 1 NE	SC	SYS/SIN	EVENT	ACE
1993-203722:10:27	18	43	Cmd Mod On	. —— - ——
1933 -203122:15:55	94	45	this is a new mes sage	CSA
993 -2 80122:01:40	94	63	U/L Handover DSS -> DSS (18.2 kN RCP)	КЈН
993 -28012 2:10:43	94	DACS2	Cmd Mod Off	KJM
993-280122:11:01	94	DACS2	คอร	CSA
1993 -260 122:24:13	94	UHC S- RT	LOS	кли
.993-280122:24:29	94	UMCS-RT	TLM Swap DSS -> DSS (SNR-)	KJM
1933 -2 80T22 <b>:26:13</b>	94	43	eos	CSA
1993-280122:43:03	94	45	HS [n SNR = *, ACC = *	c m
1993 -280 <b>123:05:52</b>	94	45	Cmd Mod Ori	кјн
993-281701:09:24	94	UMCS-RT	Cmd Mod On 42: 34m Camberra	CSA
	. = .	FUNCT ASSESSMENT OF	Coldstone. D 43: 70m Canberra	e was company promise a second

Figure 2 - OEL On-Line Logging Tool (OLOG)

## Mission Control Team operations Log

DOY: 211

Page 1 7/30/93 SYS/SIN EVENT ĴΕ TIME C 001-00:00:00 GLLTISB 993-054103:00:27 BCUI received SFOC G/W query for routing tb1[054-03:00:27] ID 993-211720:46:48 45 s/c 2-Way I d: C307 --SCPTelemetry SFDU # 25 IDP-1D = C307 SCP Telemetry 993-211 120:48:38 00c0 eb73 0079 1891 0be1 0019 0091 SCLK: 41?158945 Segment: 25 EDF Status Word 0091 MISSION MODE D=0, A=2 EDF Digital Telemetry 68 58 10 7f **7**f 40 **4**8 ef ff 9a EDF Analog Telemetry 00 0a &a 66 7b 01 3c b4 b3 11 46 7a 7b 00 b4 00 7a 78 74 7a 46 88 6e 3b e7 83 bc 83

Figure 3 - Sample 01 .OG Printout

SCP Status Word de84 from CONTROL SCP

,c006 c2b0 0001 feec 0001 8c35 13c2 cf54 0eb9 f872 0019 0008 eed8 fffc fble ffff 0522 10d8 b061 50e3 0000 0000 ffe8 0067 0032 00000000 5898 49d4 c006 ce30 0001 feec 0001 89f0 122d

CV Word c00G SCP Telemetry

DDDHH;HH;	ACTION	EVENT DESCRIPTION D		HIND RTC)	
280 21:42:1		EMBLE> ROCYET ENGINE ASSEMBLY 13 TURN ON> CATBED HEATER	`RR	3£ 13	27, 21:29:43
		P-0079 REA_CBH13_EN P-0962 REA_CBH13_TP 15 530 RED: 10 550 DEC_C			
290 21:42:1		ENGBLE> ROCKET ENGINE ASSEMBLY 15 TURN OH> CATBED HEATER	'RR	15E	275 21:29:43
		P-0001 REA_CBH15_EN P-0964 REA_CBH15_TP 15 530 RED; 10 550 DEC_C			
2.30 21:42:1		ENABLE> ROCKET ENGINE ASSEMBLY 10 TURN ON> CATBED HEATER	RR	10E	275 2, 129144
		P-0076 REA_CBH10_EN P-0959 REA_CBH10_TP 15 530 RED; 10 55 DEG_C			
280 21,42,1		ENABLE> ROCKET ENCINE ASSEMBLY 12 TURN ON> CATBED HEATER	RR	17E	275 21;29;44
		P-0078 REA_CBH12_EN P-0961 REA_CBH12_TP 15 530 RED: 10 550 DEG_C			
280 21:42:1		ENABLE > ROCKET ENGINE ASSEMBLY 14 TURN ON> CATBED HEATER	F:R	14F	275 21; 29;44
		P 3083 REA_CBH14_EN P 3963 REA_CBH14_TP 18 530 RED: 10 550 DEG_C			
283 21;42;1		ENABLE> ROCKET ENGINE ASSEMBLY 16 TURN ON> CATBED HEATER	kR	16E	275 21;29;44
		P-0002 REA_CBH16_EN P-0965 REA_CBH16_TP 15 530 RED: 10 550 DEC.C			

Figure 1 - Sequence of Events (SOE) Printout

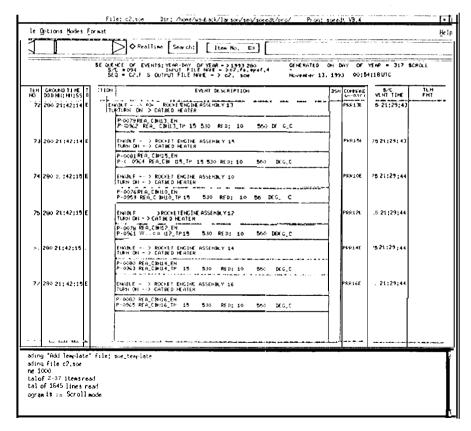


Figure 4 - Realtime Closed 1.oop SOE Tool